

# Finite Element Modeling of Stockbridge Damper and Vibration Analysis: Equivalent Cable Stiffness

Nitish Kumar Vaja, Oumar Barry, Brian DeJong

**Abstract**—Aeolian vibrations are the major cause for the failure of conductor cables. Using a Stockbridge damper reduces these vibrations and increases the life span of the conductor cable. Designing an efficient Stockbridge damper that suits the conductor cable requires a robust mathematical model with minimum assumptions. However it is not easy to analytically model the complex geometry of the messenger. Therefore an equivalent stiffness must be determined so that it can be used in the analytical model. This paper examines the bending stiffness of the cable and discusses the effect of this stiffness on the natural frequencies. The obtained equivalent stiffness compensates for the assumption of modeling the messenger as a rod. The results from the free vibration analysis of the analytical model with the equivalent stiffness is validated using the full scale finite element model of the Stockbridge damper.

**Keywords**—Equivalent stiffness, finite element model, free vibration response, Stockbridge damper.

## NOMENCLATURE

d	Diameter of messenger
E	Young's modulus
I	Area moment of inertia
J	Rotational inertia
L	Length of cable
m	Mass of messenger per unit length
M	Mass of counterweight
T	Kinetic energy
V	Potential energy
W	Transverse displacement
$E_{eq}$	Equivalent Young's modulus
$f_c$	Frequency with messenger as cable
$f_r$	Frequency with messenger as rod
$I_c$	Area moment of inertia of the cable
$I_r$	Area moment of inertia of the rod
$m_c$	Mass of messenger as a cable
$m_r$	Mass of messenger as a rod
$\omega$	Natural frequency

## I. INTRODUCTION

**P**OWER lines are often exposed to winds with speeds up to 7 m/s, which cause vortex shedding [1]. The continuous force exerted by the wind causes the conductor to vibrate at a frequency of 3 to 150 Hz. Such vibrations are characterized by small amplitudes [2] and are referred to as "Aeolian Vibration". Uncontrolled Aeolian vibrations might lead to catastrophic failure of power lines [3] as continuous vibration causes bending and tensile stresses in the conductor [4]. Stockbridge dampers are used to

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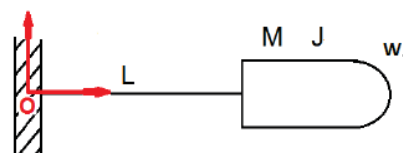


Fig. 1 Schematic of a cantilever beam with a tip mass

control these vibrations. The Stockbridge damper was first developed by George H. Stockbridge in 1925 [5]. The damping mechanism is observed as the vibrations of the conductor are transferred to the Stockbridge damper, and the energy of the conductor is imparted to the oscillating counterweights [6], [7]. The symmetric Stockbridge damper exhibits two resonant frequencies while the asymmetric Stockbridge damper has four [8], [9]. Conventional mathematical models of the Stockbridge damper assume the system to be a 2 DOF system [10], [11] with the counterweight to be a lumped mass and the messenger to be a massless beam. Other nonlinear models [12], [13] use the energy method to model the system. However the latest approach by Barry et.al considers the counterweight as a mass with rotational inertia and employs the bending stiffness of the messenger [14]. A similar mathematical model is developed in this work using Hamilton's principle to derive the governing system equations.

## II. SYSTEM DESCRIPTION

The Stockbridge damper consists of two masses connected by a messenger with a clamp in its mid-span. For mathematical simplicity, the half model of the Stockbridge damper is analyzed. This model behaves as a cantilever beam with a tip mass. The messenger is a bunch of metal strands knit together to form a cable [8]. The cable is made of galvanized steel. Grey cast iron is used for the counterweight, while aluminum alloy is used for the clamp to reduce its weight [15]. The schematic of the system is depicted in Fig. 1.

## III. EQUATION OF MOTION

The reference frame is attached to the clamp, since it is considered as the fixed end of the cantilever beam. The kinetic and potential energies of the system are given by (1) and (2).

$$T = \frac{1}{2}m \int_0^L \dot{W}^2(x,t)dx + \frac{1}{2}M\dot{W}^2(L,t) \quad (1)$$

$$V = \frac{1}{2}EI \int_0^L W''^2(x,t)dx \quad (2)$$

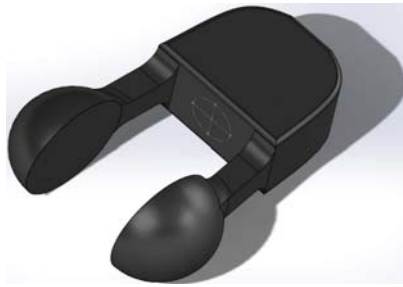


Fig. 2 Model of the counterweight



Fig. 3 Model of the cable



Fig. 4 Meshed model

Here the primes represent differentiation with respect to the axial coordinate  $x$  and dots denote the differentiation with respect to time. Using Hamilton's principle, the equation of motion of the system is obtained as the following:

$$EIW^{IV} + m\ddot{W} = 0 \quad (3)$$

Assuming the system exhibits harmonic motion,

$$W(x, t) = y(x)e^{i\omega t} \quad (4)$$

Substituting (4) into (3), the following non dimensional equation is obtained:

$$Y^{IV}(\xi) - \Omega^4 Y(\xi) = 0 \quad (5)$$

where,  $y = YL$ ,  $x = \xi L$ ,  $\Omega^4 = \frac{m\omega^2}{EI} L^4$  and the shape function of the system is written as the following:

$$Y = c_1 \sin \Omega \xi + c_2 \cos \Omega \xi + c_3 \sinh \Omega \xi + c_4 \cosh \Omega \xi \quad (6)$$

The boundary conditions are given as:

$$Y(0) = Y'(0) = 0;$$

$$Y''(\xi) = \gamma \Omega^4 Y(\xi); Y'''(\xi) = -\alpha \Omega^4 Y'(\xi)$$

where  $\gamma = \frac{J}{mL^3}$ ;  $\alpha = \frac{M}{mL}$ , the characteristic equation of the system is obtained by substituting the boundary conditions in the shape function

$$\gamma \alpha \Omega^4 (cch - 1) + \gamma \Omega^3 (sch + cch) + \alpha \Omega (sch - csh) - 1 - csh = 0 \quad (7)$$

where  $s = \sin \omega \xi$ ,  $c = \cos \omega \xi$ ,  $sh = \sinh \omega \xi$ ,  $ch = \cosh \omega \xi$ .

#### IV. FINITE ELEMENT MODELING

A half model of Stockbridge damper is developed using SolidWorks. The counterweight, messenger, clamp, and bushing are modeled independently and the assembly is meshed. Two models are developed: one with the messenger as a stranded cable and the other with the messenger as a rod. The stranded messenger is modeled as a 1x19 cable, and it is ensured that there is frictionless contact between the surfaces of individual strands [8]. The messenger is made of galvanized steel with Young's modulus  $2.00E+11 \text{ N/m}^2$ , Poisson's ratio 0.29 and mass density  $7870 \text{ kg/m}^3$ . The Young's modulus of the counterweight is  $6.62E+10 \text{ N/m}^2$  and its Poisson's ratio was 0.27 with a mass density  $7200 \text{ kg/m}^3$ .

The modeled parts are assembled and meshed using a curvature based mesh with four Jacobian points. The

TABLE I  
 FIRST 10 FREQUENCIES OF TWO MODELS WITH  $d = 5.04 \text{ mm}$

Mode No.	Strand(rad/s)	Rod(rad/s)	Ratio( $f_c/f_r$ )
1	18.801	3.817	0.147547
2	20.204	3.938	0.147693
3	56.103	11.480	0.152579
4	78.952	15.324	0.150305
5	89.126	17.169	0.150216
6	487.920	266.230	0.405206
7	902.770	266.370	0.274766
8	984.950	266.400	0.247847
9	1093.100	266.420	0.205202
10	1538.000	266.430	0.163470

maximum element size is  $1.09424 \text{ mm}$  while the minimum element size is  $0.218848 \text{ mm}$ . The total nodes are 1,712,657 and total elements are 1,111,750.

The rotational moment of inertia of the counterweight is  $0.002087 \text{ kgm}^2$  and its mass is  $1.1298 \text{ kg}$ . The length of the messenger is  $0.130 \text{ m}$ . A frequency response analysis is conducted on both of the models, and the results are tabulated in Table I.

#### V. RESULTS

The frequency ratio for each model is calculated. In the first case (Case 1), the diameter of the messenger is  $5.04 \text{ mm}$ . The other four cases are obtained by increasing the diameter of the messenger by  $2.5 \text{ mm}$  for each case. All the other parameters (counterweight mass and messenger length) are kept the same. The average of the ratios are presented in Table II. These ratios were used to conduct regression analysis and to calculate the equivalent Young's modulus of the stranded cable.

The frequency ratios are plotted against the diameter of the messenger in Fig. 9. Using regression analysis, a relation between the frequency ratio and the diameter of the messenger is obtained as:

$$f_{ratio} = -0.0005d^2 + 0.0067d + 0.1222 \quad (8)$$

The area moment of inertia of a circular cross section is simple to calculate, but the area moment of inertia of the cable with complex cross sectional area is more difficult to calculate.

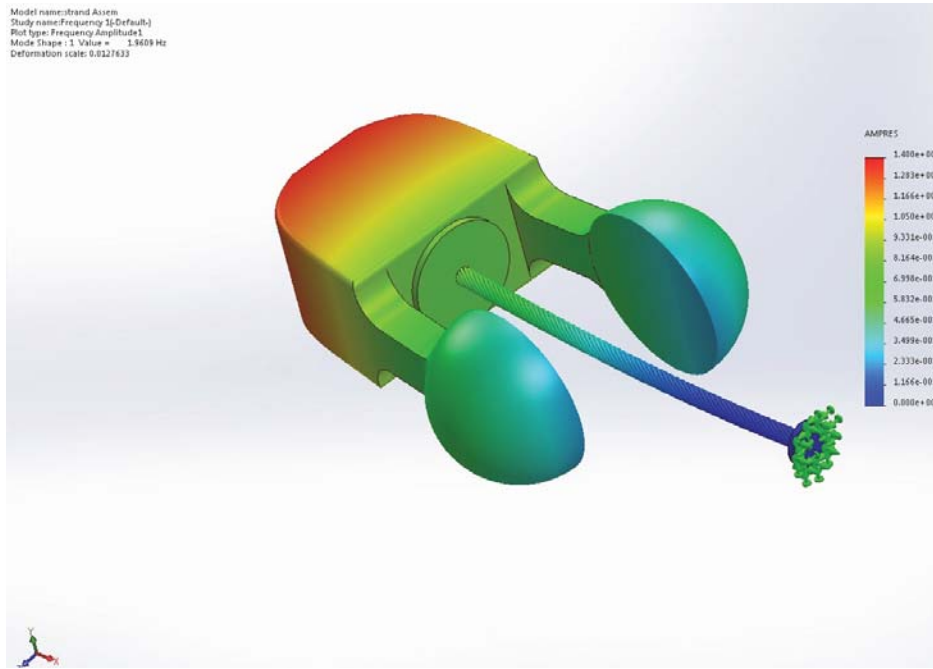


Fig. 5 First mode shape:  $f = 1.96\text{Hz}$

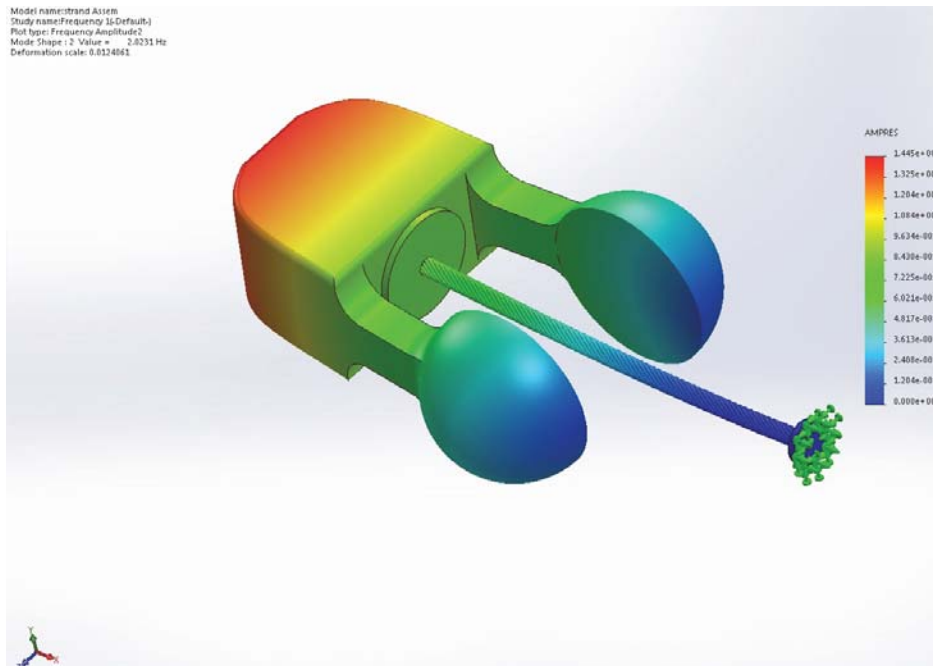


Fig. 6 Second mode shape:  $f = 2.02\text{Hz}$

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The moment of inertia of cable and rod of different diameters is presented in Table III. It is observed that the moment of inertia of the cable was proportional to moment of inertia of the rod and the relation is presented by (9).

$$0.71I_r = I_c \quad (9)$$

The equivalent Young's modulus is obtained in terms of the

diameter and it is presented in (10).

$$E_{eq} = 3.24 \frac{E}{I_{ratio}} (-0.0005d^2 + 0.0067d + 0.1222)^2 \quad (10)$$

where  $I_{ratio}$  is the ratio of  $I_c$  and  $I_r$ . This ratio is equal to 0.71 as given in (9).  $E$  is the Young's modulus of the material ( $2E+11 \text{ N/m}^2$ ). The deduced equivalent stiffness is used in the analytical model to calculate the natural frequencies, which are then compared with the numerical model. Fig. 10 shows that

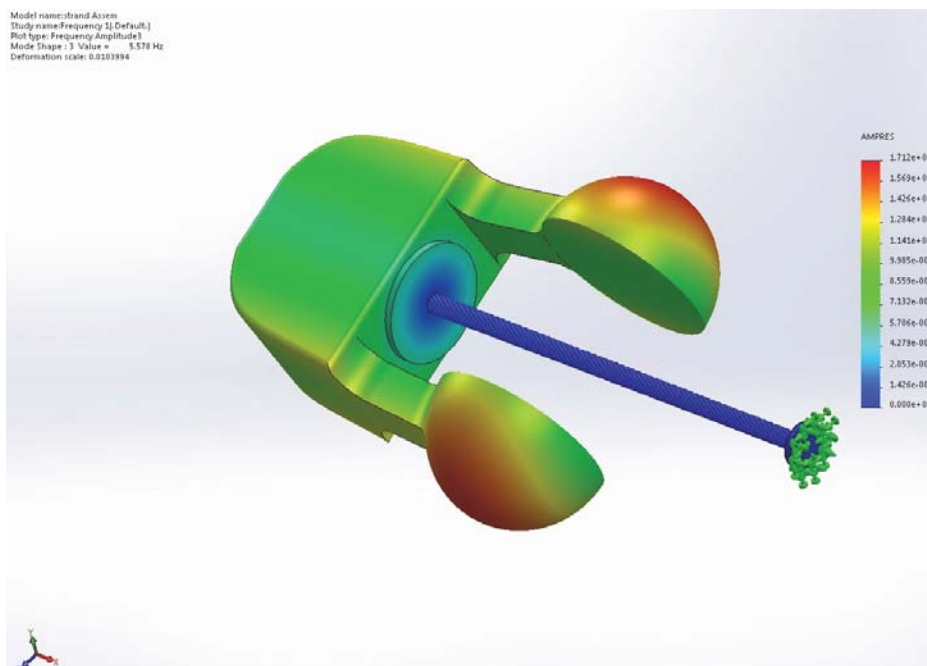


Fig. 7 Third mode shape:  $f = 5.578\text{Hz}$

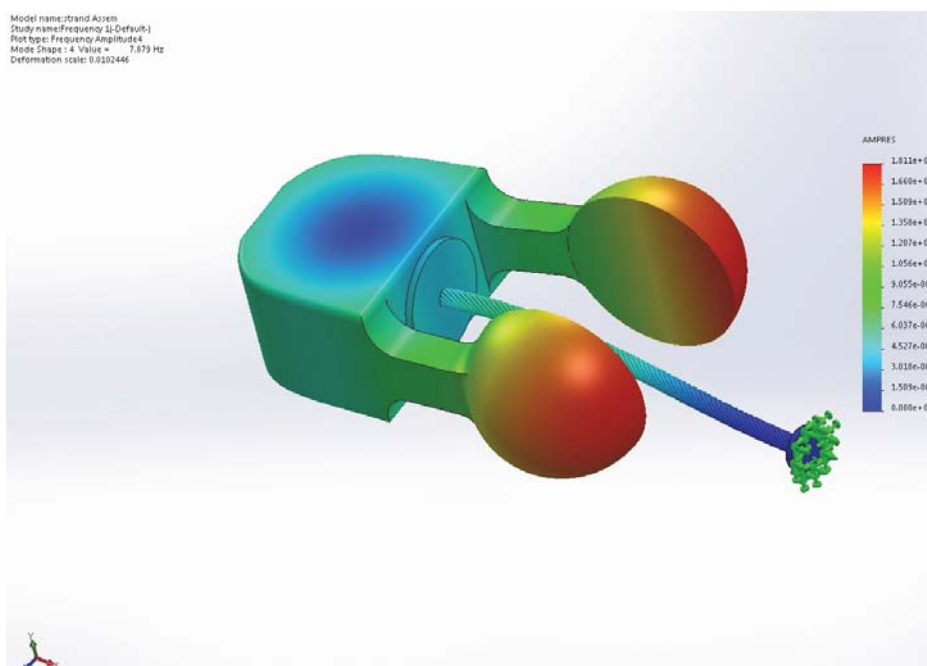


Fig. 8 Fourth mode shape:  $f = 7.879\text{Hz}$

the results from the analytical model are in good agreement with the results from the numerical model.

## VI. CONCLUSION

The observations from this paper will enable designers to obtain the equivalent stiffness of a stranded cable with reference to a rod with equal diameter. This would save a lot of energy that is put into experiments for determining messenger bending stiffness, which is both uneconomical and

time consuming. The equivalent stiffness could be used in the conventional model and the natural frequencies of the Stockbridge damper could be calculated precisely.

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TABLE II  
 AVERAGE FREQUENCIES OF TWO MODELS AND THEIR RESPECTIVE RATIOS

Diameter(mm)	Average Ratio ( $f_c/f_r$ )
5.04	0.143822
6.29	0.145150
7.54	0.150216
8.79	0.144408
10.04	0.145914
11.29	0.139029
12.54	0.133327
13.79	0.124206
15.04	0.122413

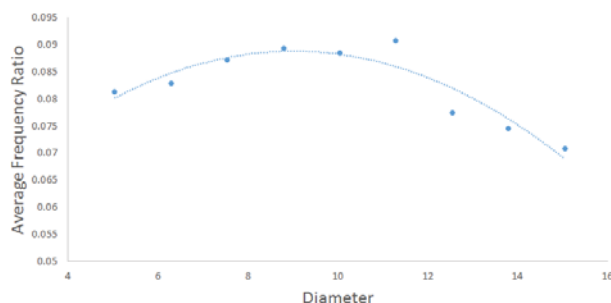


Fig. 9 Regression analysis

TABLE III  
 COMPARISON OF MOMENT OF INERTIA OF MESSENGER

Case	$I_r(m^4)$	$I_c(m^4)$	Ratio ( $I_c/I_r$ )
Case 1	3.16E-11	2.26E-11	7.14E-01
Case 2	1.58E-10	1.14E-10	7.22E-01
Case 3	4.98E-10	3.58E-10	7.19E-01
Case 4	1.21E-10	8.71E-11	7.18E-01
Case 5	2.51E-09	1.80E-09	7.17E-01

Analytical frequency vs Numerical frequency

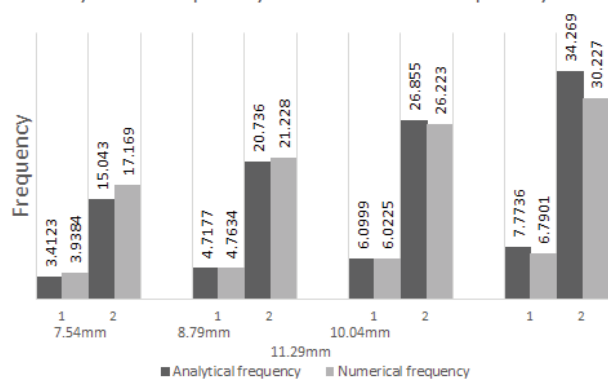


Fig. 10 Comparison between the first two modes for different diameters

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